

GFA AD 575

(ISSUE 1)

GFA AIRWORTHINESS DIRECTIVE

TYPE AFFECTED:

Ventus 2c, serial numbers 1 to 76.

SUBJECT:

CG (belly) release, installation position.

BACKGROUND:

Cases have occurred of failure to back-release. The installation position of the Tost release must be changed in accordance with manufacturer's instructions and the maintenance manual amended.

DOCUMENTATION:

Schempp-Hirth Technical Note No 349-26, including maintenance

manual amendment, forms part of this AD.

ACTION REQUIRED:

Carry out actions specified in Schempp-Hirth Technical Note No

349-26 and amend maintenance manual as specified.

WEIGHT AND BALANCE: Not affected.

IMPLEMENTATION:

At or before the next annual inspection, but in any case not later

than 31 December 2002.

COMPLIANCE:

The requirements of this GFA Airworthiness Directive are

mandatory. This Directive is issued pursuant to the Rules and

Regulations of the Gliding Federation of Australia.

SIGNED:

For and on behalf of:

THE GLIDING FEDERATION
OF AUSTRALIA

GFA AD 575

ISSUE: 1

SENIOR TECHNICAL OFFICER AIRWORTHINESS

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9 May 2002

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SCHEMPP-HIRTH Flugzeugbau GmbH.	Technical Note No. 349 - 26	Page No:	01
Kirchheim/Teck	No. 360 - 18	No of pages.:	02

SUBJECT:

C.G. release, installation position

AFFECTED:

Sailplane Ventus-2c (TC-No.: 349)

S/N 1 through 76

Sailplane Discus-2b (TC-No.: 360)

S/N 1 through 133

URGENCY:

At the occasion of the next annual inspection

but not later than December 31, 2002

REASON:

On two occasions and on two different sailplanes the winch cable

did not release automatically from the Tost "Europa G 88 Safety

Release".

The installation position of the release will, therefore, be changed

and the release mounted in the lowest hole.

For sailplanes where the release is mounted directly on to the bottom of the fuselage with the special solid-steel ring protruding,

the release cover must also be modified.

ACTIONS:

1. Only for sailplanes equipped with a C.G. release or when retrofitting the C.G. release.

The modifications were carried out in accordance with the following drawings and the instructions in the appendix.

Drawing number	Title
S08 RB 850	Installation of the C.G. release
S11 RB 842	Installation of the countersunk C.G. release

Remark: Cut outs of the drawings are given in the Maintenance Manuals, see "ACTIONS" item 2.

SCHEMPP-HIRTH Flugzeugbau GmbH. Kirchheim/Teck		No. 3	cal Note 49 - 26 60 - 18		Page No.: No of pages.:	02	
					1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		
ACTIONS (ctd.)	2. Revision	ns of the Mai	ntenance Ma	anual			
	 Revisions of the Maintenance Manual (revised pages dated January 2002) 						
		variant					
		Discus-2b	Ventus-2c	Title			
	page	0.1.3	0.1.3	Record	of revisions		
	page	5.2.4	5.2.3	Installati	ation of C/G release		
	<u> </u>	<u>. </u>	<u> </u>	<u> </u>	·		
	List of e	ffective page	s to be ame	nded by h	and.		
	Screwe	r sailplanes	able connecti	ion at the	C.G. release	<u>e</u>	
	release	cover part ac	ditionally				
		Ventus-2c, S/N 1 through 62					
		Discus-2b,	S/N 1 thr	ough 55 a	and 57 through (64	
<u>WEIGHT:</u>	No alteration	on					
, , , , , , , , , , , , , , , , , , ,			•				
C/G POSITITION:	No alteration	on					
<u>NOTE:</u>	This action and to be on inspector.	s may be ac checked and	complished t entered in th	oy an exp se log boo	erienced persor ok by a licenced	ı	
Kirchheim/Teck, Decemb	per 21 2001						
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The translation into English has been done by best knowledge and judgement.

Issued: YU (Treiber) SCHEMPP-HIRTH Flugzeugbau GmbH. Kirchheim/Teck

Appendix to Technical Note No. 349-26 No. 360-18

Page No:

01

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- 1. Remove seat pan
- 2. Remove release cover
- 3. Remove cable connection to release
- 4. Release attachment
 - a) Loosen aft screw (number 5)
 - b) Remove forward screw. Rotate release upwards and attach release again in the lowest hole (number 4)
 - c) Tighten again aft screw (number 5)
- 5. Attach cable connection with the additional fittings, see photo page 03.
- Increase the cut out in the fuselage bottom for the release with the aid of the templates, see page 02.
 The cut out should be increased so far that with the cable ring engaged and pulled forward the pair of cable rings support against the automatic release device.
- Modify the release cover only for Ventus-2c, S/N 1 through 62 Discus-2b, S/N 1 through 55 and 57 through 64
 - a) Glue the additional release cover on top of the present cover
 - b) After curing of the additional release cover abrade the inner part of the present release cover

<u>Remark</u>

The size of the release cover for the **countersunk** release is big enough to house the release in the modified installation position and must not be changed, see

Ventus-2c, S/N 63 through 76 Discus-2b, S/N 56, 65 through 133

- 8. After checking the cable connection and the release operation mount the release cover and check again the release for free operation
- 9. Install seat pan

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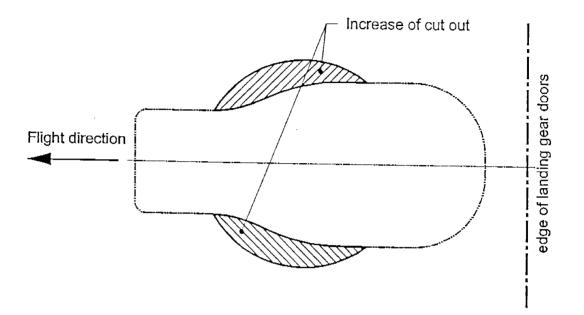
Appendix to Technical Note No. 349-26 No. 360-18

Page No:

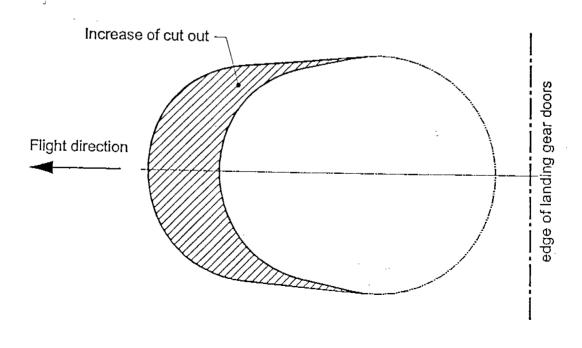
02

No of pages.: 03

Fuselage cut out for the not countersunk release installation



Fuselage cut out for countersunk release installation



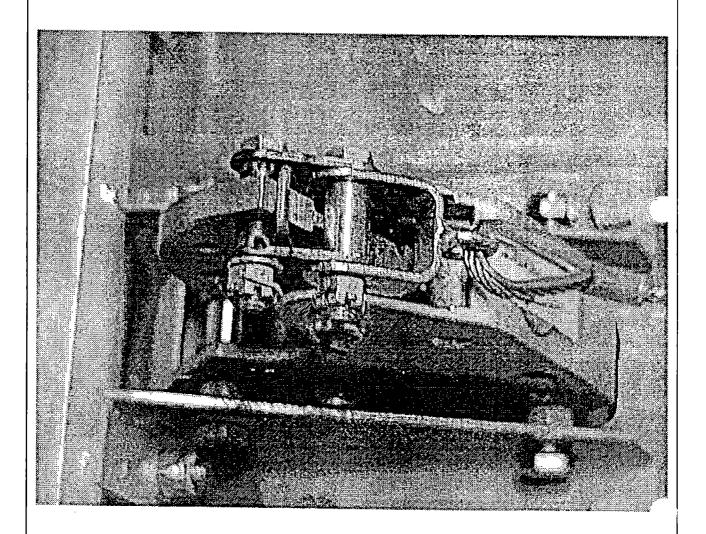
SCHEMPP-HIRTH Flugzeugbau GmbH. Kirchheim/Teck Appendix to Technical Note No. 349-26 No. 360-18

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03

Cable elongation for release operation



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Ventus-2c

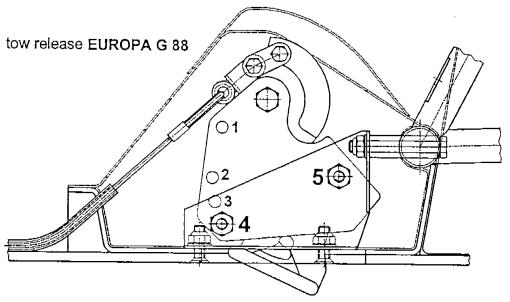
WARTUNGSHANDBUCH / MAINTENANCE MANUAL

0.1 Erfassung der Berichtigungen / Record of Revisions

Lfd.Nr. Rev.No.	Benennung Reference	Seite Page	Datum Date
9	Technische Mitteilung Nr. 349-26 Einbauposition der Schwerpunktkupplung Werk-Nr. 1 bis 76 Technical Note No. 349-26 C/G tow relelase, mounting position S/N 1 through 76	5.2.3	Januar 2002
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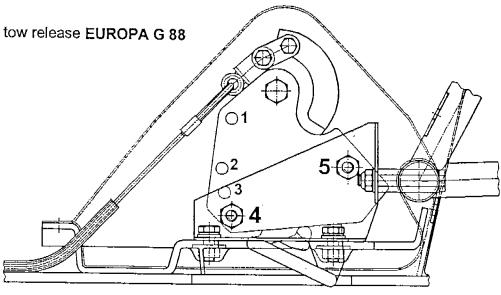
MB: Modification Bulletin — Änderungsblatt TN: Technical Note — Technische Mitteilung

EXTRACTED FROM DRAWING NO. S08 RB 850



to be attached by bolts No. 4 and 5

EXTRACTED FROM DRAWING NO. S11 RB 842



to be attached by bolts No. 4 and 5