



## **GFA AIRWORTHINESS DIRECTIVE**

## **CANCELLED 29.03.2017** REFER TYPE CERTIFICATE HOLDERS CURRENT DATA

| TYPE AFFECTE   | D: All ASH 25  | All ASH 25 sailplanes and ASH 25 E motorgliders.  |                    |  |
|--|--|---|--------------------|--|
| SUBJECT:   | 1.) Increase   | e of the service life from 6000 to  | 0 12000 hours.     |  |
|  | 2.) Inspectio  | on of elevator control linkage.   |                    |  |
|  | 3.) Addition   | nal safety device for the landing   | gear rear bolts.   |  |
|  | 4.) Exchange   | ge of pages in the Flight and Ma  | intenance Manuals. |  |
| BACKGROUND   | on fibre con<br>service life<br>structural of<br>examine the<br>of 12000 he<br>of each ind<br>inspection p | 1.) <i>Increase of service life from 6000 to 12000 hours:</i> Fatigue tests on fibre composite wings and wing spars have demonstrated that a service life expectancy of 12000 hours can be reached for these structural components. As these fatigue test programs did not examine the entire aircraft made of fibre composite, the service life of 12000 hours can be granted only if the long-term airworthiness of each individual aircraft is demonstrated in a special multi-stage inspection program (over and above the mandatory annual C of A inspections) for the purpose of increasing the service life. |                    |  |
| <ul> <li>2.) <i>Inspection of the elevator control linkage:</i> In the case of two ASH 25, damages to the elevator control linkage have been overlooked after inadequately accomplished repairs in the areas of the landing gear and of the fin. These damages due to overstressing the elevator control linkage can lead to reduced control stiffness. The decrease of elevator frequency caused by this can in turn when being initiated accordingly, lead to a frequency coupling with the horizontal stabiliser and thus lead to flutter.</li> <li>3.) <i>Additional safety device for the landing gear rear bolts:</i> The</li> </ul> |  |   |                    |  |
| two bolted connections of the landing gear H-strut to the fittings<br>the rear landing gear bulkhead have come loose in single cases.  |  |   |                    |  |
| these bolted connections are not covered with screw safety varnish         SIGNED:       T Geiger         For and on behalf of:  |  |   |                    |  |
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|-----------------------------------|--|--|----------------------------|--|
|                                   | must be ad   | the bolted connection on each side must be undone and a lockplate<br>must be added which must be bent upwards after retightening of the<br>bolted connection.  |                            |  |
| DOCUMENTAT                        | Technical<br>drawings a                                | Manufacturer's ASH 25 Technical Note No. 14 and ASH 25 E<br>Technical Note No. 12 (attached to this AD). Further material,<br>drawings and manual pages can be obtained from the Australian<br>Schleicher agent or Alexander Schleicher GmbH & Co<br>Segelflugzeugbau<br>PO Box 60<br>36161 Poppenhausen<br>Germany<br>Ph. +49 6658 890; Fax +49 6658 8940                   |                            |  |
| ACTION REQUI                      | after the "<br>25" has be<br>elevator co<br>landing ge | 1.) In accordance with the Technical Note: Increase of service life<br>after the "Inspection Program To Increase The Service Life ASH<br>25" has been done. 2.) Inspection, and if necessary repair, of the<br>elevator control linkage. 3.) Installation of a lockplate for the<br>landing gear rear bolts. 4.) Exchange of pages in the Flight and<br>Maintenance Manuals. |                            |  |
| WEIGHT AND BALANCE: Not affected. |  |  |                            |  |
| IMPLEMENTAT                       | annual ins   | The works 1.) - 3.) must be carried out by an inspector rated for annual inspections (Form 2 inspector). The exchange of manual pages 4.) can be done by the owners.   |                            |  |
| COMPLIANCE:                       | 1.) Increase   | e of service life: before reaching   | g 6000 flight hours.       |  |
|                                   | · •  | on of the elevator control linka<br>repair in the landing gear or the  |                            |  |
|                                   |  | onal safety device for the land<br>next Form2 inspection.  | ing gear rear bolts: at or |  |
|                                   | 4.) Exchar<br>Before 31                                | nge of pages in the Flight and<br>July 1999.   | d Maintenance Manuals:     |  |
| ATTACHMENTS                       | S: Schleicher  | Schleicher Technical Note ASH 25 No. 14 and ASH 25 E No. 12.   |                            |  |