GLIDING FEDERATION OF AUSTRALIA

AD 379

(ISSUE 2)

AIRWORTHINESS DIRECTIVE

SUBJECT:

Manual revisions and inspection of Hotellier couplings

TYPE AFFECTED:

DG-400

BACKGROUND:

This AD is to ensure compliance with Glaser-Dirks Technical Note No 826/24 attached. The manual amendment pages are available

from the GFA secretariat on request.

DOCUMENTATION:

Glaser-Dirks Technical Note No 826/24.

ACTION REQUIRED:

Before 30 October, 1991 amend the flight and service manual as

detailed in Technical Note No 826/24.

Before next flight inspect the Hotellier couplings as detailed in Technical Note No 826/24. The minimum force (preload) to move the sliding latch is approximately 600 g. The force when the latch is fully home should be around 1000 g. Faulty latches have been measured with zero preload and a maximum load of 300 g. Ensure

that each coupling is clean before performing this test.

IMPLEMENTATION:

Manual amendments may be incorporated by the owner.

Inspection and possible replacement of the Hotellier couplings must be carried out by persons rated Inspection for Issue of

Certificates of Airworthiness, any type.

WEIGHT AND BALANCE: No effect

COMPLIANCE:

The requirements of this Airworthiness Directive are mandatory. This Directive is issued pursuant to the Australian Civil aviation

Regulations under the delegated authority of the Civil Aviation

Authority (CEO42/90).

SIGNED:

CHIEF TECHNICAL OFFICER AIRWORTHINESS

For and on behalf of:

THE GLIDING FEDERATION
OF AUSTRALIA

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Glaser-Dirks Flugzeugbau GmbH Im Scholengerten 19-20, 7520 Bruchsel 4 Telefon 0.7257/09-0, Telex 7822410 GLDQ LBA anerkannter Horstellungsbetrieb 10:25 LBA anerkannter Luttahrtbuchnischer Zeiten 8A:279

Technical note No. 826/24

SUBJECT : Manuals DG-400

EFFECTIVITY: Powered sailplane DG-400 from serial no. 4-1 on

ACCOMPLISHMENT: Instruction 1 until 30.October 1991.
Instruction 2 prior to next take off.

REASON : 1. Manual revision

2. The Hotellier control quick connects are subject to exceptional wear in a motorglider and should be inspected regularly. Especially a damaged spring of the sliding latch may cause disconnection of the control connection.

INSTRUCTIONS: 1. Exchange the following pages of the manuals to the new pages issued August 1991 and add some new pages.

The revisons are indicated at the right margin. Please study all revisions.

page change

flight manual

0.1 TN 826/24	
Contont	
2 content	
2a content	
fuel-two stroke oil specifications tow release versions	
28 check fuel filter	
37 Wrong data corrected	
43 flight at birth	
flight at high altitude and at low temper check spring force of Hotellian	atures
check spring force of Hotellier connects	
dittiditions or remains	
maintenance instructions	
fuel lines kinked?	

maintenance manual

0.2	TN 826/24
1	content
2	content
2a	content
3 3a 14 15 16 21 26	fuel lines at the engine life time max. 3 years new instructions for tow release SH72 tightening torque bolt 4M31/1 brake of retraction-extension motor vacuum pump note starter press button inspection of Matter.
32	inspection of Hotellier quick connects
33 34	Engine servicing schedule, fuel filter replacement - no cleaning check of starter ring gear and hinge pins fuel lines at the engine replacement after max. 3

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Technical note No. 826/24

rage 4 Irom 2

checking the ignition unit 46f,g ignition unit trouble shooting 46h, i, j

46k, 1, m removal of magneto flywheel and starter gear

typing errors

diagram 6 powerplant-screwed connection of rear propeller mountin bolts

EOD Dräger oxygen system

4EPELT(3) ELT Pointer

IM 10.01A maintenance instructions for Hotellier connects

Repair manual

0.1 TN 826/24 4/2 content

10 repairing outer skin of sandwich panel

INSTRUCTIONS

: 2a)Check the spring force of the sliding latches of all Hotellier control quick connects. Don't replace or fix damaged or kinked springs. Exchange the complete quick connect prior to

next take off in such cases. b) Check if spring pins are installed. If not, fasten spring pins to each quick connect with a thin strong cord.

Manual pages see above

2 spare spring pins 500 30 771

WEIGHT AND

MATERIAL.

BALANCE

REMARKS Instructions may be executed by the owner and

are to be entered in the aircraft logs.

Bruchsal 4, date 2.8.1991

Wilhelm Od

Author : Dipl.-Ing. W.Dirks

A. hange

LBA - approved:

The German original of this TN has been approved by LBA under the date of 91/00/12 and is signed by Mr. Fendt. The trans- i.v. lation into English has been done by best knowledge and judgement. In any case of doubt the German original is authoritative.

Type certification inspector: Dipl.-Ing. A. Lange